

**Original Research Article****The Influence of Homogenisation Treatment on Aging Response of 6063****Aluminium Alloy****ABSTRACT**

This paper reports the effect of homogenisation treatment on T6 tempering of 6063 aluminium alloy. Wrought 6063 aluminium sample was machined into tensile and impact tests specimens. Samples were also cut for hardness and metallographic works. These samples were divided into two groups; group I samples were homogenised at 580 °C for 2, 2.5, 3 and 3.5 hours respectively prior to T6 temper while group II samples were T6 tempered without prior homogenisation. The as-received sample as well as the heat treated samples was subjected to tensile, impact and hardness tests and the evolved microstructures was characterised using a scanning electron microscope equipped with energy dispersive spectrometer. The results show improvement in the mechanical properties for those samples homogenised prior to aging as compared to conventionally aged samples and there was also an unusual combination of mechanical properties in terms of ductility, toughness and strength. The resulting microstructures shows the presence of rod-like phases in the as-received and T6 tempered samples while group II samples contain spherical precipitates. The overall result showed that prior homogenisation can prevent the usual concomitant decrease in ductility and toughness of T6 tempered 6063 aluminium alloy.

**Keywords:** *Homogenisation, Age hardening, T6 temper, 6063 aluminium alloy and Wrought*

**1.0 INTRODUCTION**

Aluminium alloy 6063 and 6061 has been identified as a marine grade alloys because to their excellent corrosion resistance in marine environments. The high strength-to-weight ratio of these alloys has made them to be very attractive to aviation and automobile industries where there is

25 high demand for light materials to increase the load carrying capacity and reduce fuel  
26 consumption. The 6063 alloy seems more prominent than the 6061 because of its excellent  
27 extrudability, excellent corrosion resistance, weldability and moderate strength and other  
28 structural applications.

29 High strength aluminium alloys are usually chosen for automobile and aircraft constructions  
30 because of their high strength-to-weight ratio and stiffness, which are derived from precipitation  
31 hardening. However, high strength aluminium alloys have poor resistance to stress corrosion  
32 cracking (SCC), particularly when they are at near peak strength condition [1, 2, 3]. Precipitation  
33 hardening is directly responsible for the Stress Corrosion Cracking susceptibility of high strength  
34 aluminium alloys. This high susceptibility of 7075 aluminium for example, to corrosion  
35 especially in marine environment has shifted the attention of researchers to 6063 aluminium.

36 Aluminium alloy 7075, exhibits superior strengths (over 1.5 times that of the marine grade  
37 alloys) but is much more susceptible to corrosion [4]. This alloy sees heavy use in the aircraft  
38 industry where the environment is typically mild and aluminium corrosion is not likely to occur.  
39 Even though it is a high performance material in the aircraft industry, it would perform poorly in  
40 marine environments [4].

41 This uniqueness of 6063 aluminium alloy among other aluminium alloys demands special  
42 attention and this necessitates the need for further improvement in its mechanical properties for  
43 better performance in service, hence this study. In previous investigations on homogenisation of  
44 aluminium alloys, attention has always been on billets homogenisation prior to extrusion [5, 6,  
45 7]. Several authors studied the influence of the cooling rate after homogenisation on the alloy  
46 microstructure. Zajac *et al.* [8], Nowotnik and Sieniawski [9] studied the influence of the cooling  
47 rate on the final mechanical properties of 6063, 6082, 6005 alloys. Reiso [10] studied the

48 influence of the cooling rate on the extrusion speed for various chemical compositions of Al-Mg-  
49 Si alloys. Birol [11] studied the microstructure evolution of the 6063 alloy during  
50 homogenisation for various thermal cycles. Cai *et al.* [12] studied the Mg<sub>2</sub>Si dissolution during  
51 homogenisation through electrical resistivity measurements and the distribution of the alloying  
52 elements with electron microprobe measurements for the 6061, 6069 alloys. Finally, Usta *et al.*  
53 [13] studied the dissolution/coarsening kinetics of the Mg<sub>2</sub>Si particles during reheating of the  
54 homogenised material. This present work investigates the influence of homogenisation treatment  
55 on the precipitation hardening of 6063 aluminium alloy.

## 56 **2.0 MATERIALS AND METHODS**

57 The 6063 aluminium alloy used for this study was sourced from Nigeria Aluminium  
58 (NIGALEX), Lagos. Wrought 6063 aluminium alloy of 15 mm diameter and 1000 mm length  
59 were given. The extrusion process entails direct chill casting of the ingots, homogenisation of the  
60 ingots, preheating of the homogenised ingots, followed by hot extrusion. But the full potentials  
61 of homogenisation could not be attained in the homogenisation treatment of the ingots because  
62 this will lead to increment in the amount of stress required for extrusion. The elemental  
63 composition of wrought 6063 aluminium alloy used is presented in Table 1. Standard mechanical  
64 test samples were machined from this rod for tensile and impact tests. Samples were also cut for  
65 microhardness and metallographic works. These samples were divided into two groups; group I  
66 samples were homogenised at 580 °C for 2, 2.5, 3 and 3.5 hours and air cooled, solution treated  
67 at 540 °C for 4 hours, quenched in water and artificially aged at 185 °C for 5 hours. The group II  
68 samples were solution treated at 540 °C for 4 hours, quenched in water and artificially aged at  
69 185 °C for 5 hours without prior homogenisation. The heat treated and as-received samples were  
70 subjected to tensile, impact and microhardness tests. Tensile test was carried out in accordance

71 with British Standard BSEN 10002-1 [14] at room temperature with a crosshead speed of 5  
 72 mm/min using an Instron 3369 electromechanical testing machine. The proof stress, ultimate  
 73 tensile strength, percentage elongation and modulus of elasticity values were calculated from  
 74 Stress – Strain diagrams obtained. Impact testing of all these specimens was conducted in  
 75 accordance with ASTM Standard E 602-91 [15]. Three samples were tested from each heat-  
 76 treated condition and as-received samples. The tests were carried out using Izod impact test  
 77 method on a Houndsfield balance impact-testing machine. An average value from three tests was  
 78 taken are recorded. Microhardness testing was done using the LECO ASTM E384 microhardness  
 79 tester. The tests were performed on the six etched samples used for the scanning electron  
 80 microscopy. The microhardness test was carried out using a test load of 490.3 mN and dwell  
 81 time of 10s. This test was done at three different points on each sample and the average hardness  
 82 value reported. The samples for scanning electron microscopy (SEM) in each of the six  
 83 conditions were grinded with emery grit papers and polished to 0.5 micron finish followed by  
 84 etching with Keller’s solution (1.0 ml HF, 1.5 ml HCl, 2.5 ml HNO<sub>3</sub> and 95.0 ml H<sub>2</sub>O) [16].

85 **Table 1. Elemental Composition of the 6063 Alloy used**

Element	Si	Mg	Fe	Cr	Ti	Mn	Zn	Cu	Al
wt. %	0.53	0.43	0.13	0.14	0.02	0.04	0.01	0.17	Balance

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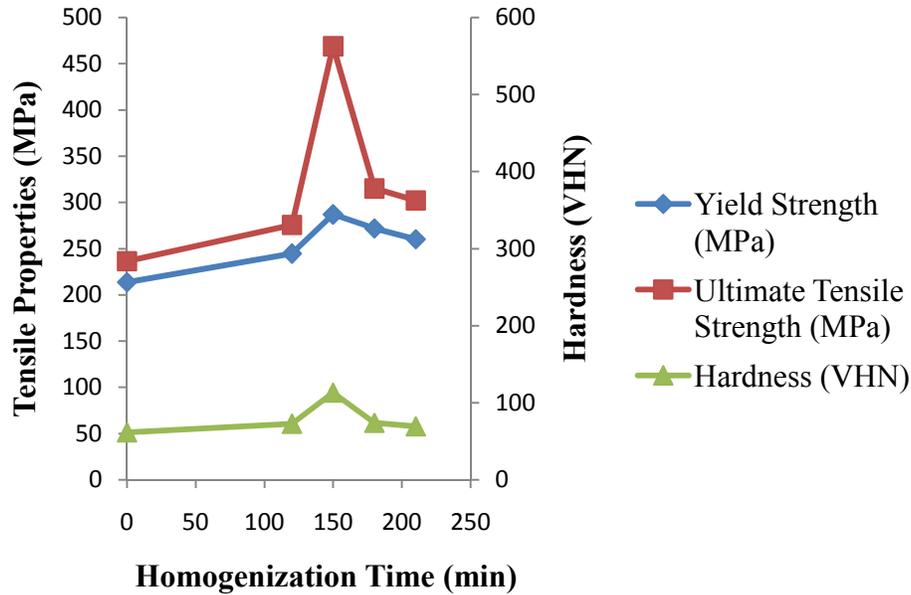
### 87 **3.0 RESULTS AND DISCUSSION**

88 The need for further homogenisation is obvious in the microstructures of the as-received sample  
 89 in Figure 2 and T6 tempered sample in Figure 3. The direct chill casting of ingot is associated  
 90 with the evolution of plate-like phases that have deleterious effects on the mechanical properties

91 of the alloy. These badly shaped phases must be completely eliminated via homogenisation prior  
92 to aging.

93 The yield strength values for the samples increased significantly after homogenisation. This is  
94 seen in Fig. 1 where there was an increase from 210 MPa for no homogenisation to 240 MPa  
95 after homogenisation for 2 hours. However, maximum value was obtained after homogenisation  
96 for 2.5 hours beyond which it decreased slightly. The same trend, as for the yield strength was  
97 observed for the ultimate tensile strength, as seen in Fig. 1. In these cases, homogenisation prior  
98 to solution treatment has been found to be very necessary. The significant increase has been  
99 found to be due to: removal of deleterious intermetallic phases and structures which are hard to  
100 remove by solution treatment only, enriching the solid solution matrix with solute atoms for  
101 solution strengthening and release of the solute atoms for subsequent formation of favorable and  
102 coherent precipitates.

103 Figures 2 and 3 are the microstructures of the as-received and samples solution treated and aged  
104 but without prior homogenisation. The presence of incoherent and elongated intermetallic phases  
105 in these two Figures is responsible for low yield and ultimate tensile strengths values. The  
106 sudden increase in the mechanical properties, as shown in Fig. 1 above, after 2 hours  
107 homogenisation was as a result of complete spheroidization of plate-like phases present in the  
108 alloy matrix and the formation of suitable heterogeneous nucleants (dispersoids) which enhances  
109 the quench sensitivity and precipitates formation during subsequent aging.



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111 **Fig. 1. Variations of yield strength, ultimate tensile strength and hardness with homogenisation time for T6**  
 112 **tempered and homogenized prior to T6 tempered 6063 aluminium alloy.**

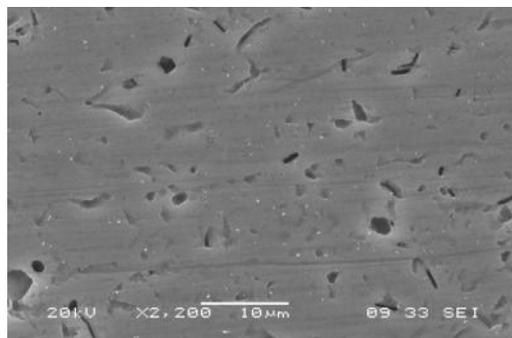
113 This increases the density of precipitates formed and thereby strengthening the alloy by exerting  
 114 great barrier to dislocation movement. The sudden drop at about 30 minutes latter could be  
 115 attributed to the formation of incoherent precipitates as a result of excessive homopgenisation.

116 The optimum values of yield and ultimate tensile strength (287 and 470 MPa) were obtained at  
 117 2.5 hours homogenisation prior to aging. This means that complete homogenisation treatment  
 118 was achieved at 2.5 hours during which there was removal of segregations, spheroidisation of  
 119 dispersoids and transformation of  $\beta$ -Al-Fe-Si to  $\alpha$ -Al-Fe-Si for the formation of coherent  
 120 precipitates during aging and this is in agreement with the results of [5, 17, 18] whose results  
 121 show that no more than 2.5 hours is required for complete homogenisation of 6063 aluminium  
 122 alloy. This structure is seen in Fig. 4, where complete spheroidisation took place.

123 However, the reduction in mechanical properties after 2.5 hours homogenisation is likely to be  
 124 due to formation of non-coherent precipitates caused by excessive dissolution of solute atoms

125 during extensive homogenisation. This has resulted in a condition for over aging. This is evident  
126 in the amounts of precipitates in Figures 5 and 6 as compared with Fig. 4.

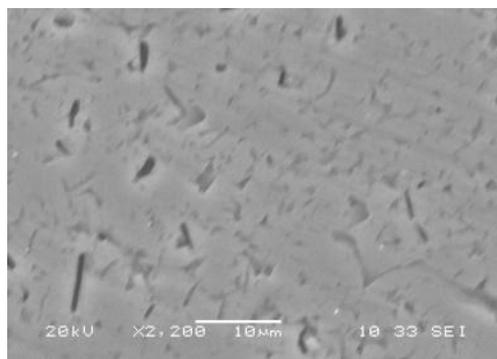
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**Fig. 2. SEM micrograph of the as-received 6063 aluminium alloy used for this study.**

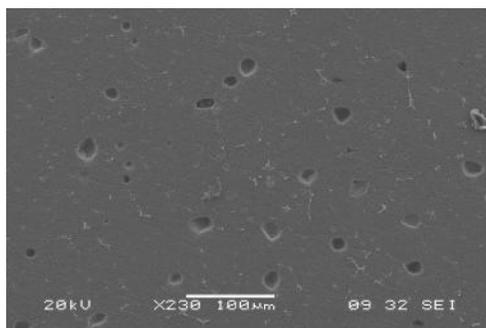
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**Fig. 3. SEM micrograph of T6 Tempered 6063 aluminium alloy**

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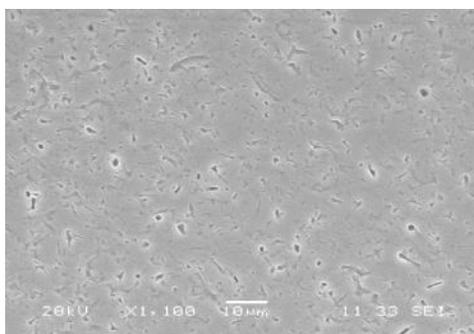


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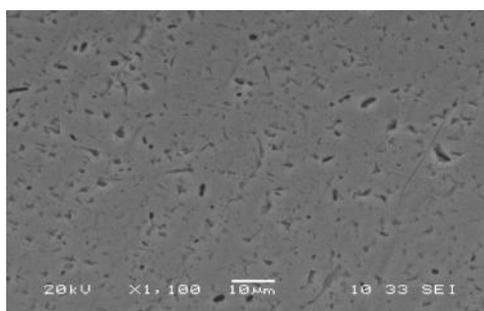
**Fig. 4. SEM micrograph of 6063 aluminium alloy homogenised at 580 °C for 2.5 hours and T6 tempered.**

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135 **Fig. 5. SEM micrographs of 6063 aluminium alloy homogenised at 580 °C for 3 hours and T6 tempered.**

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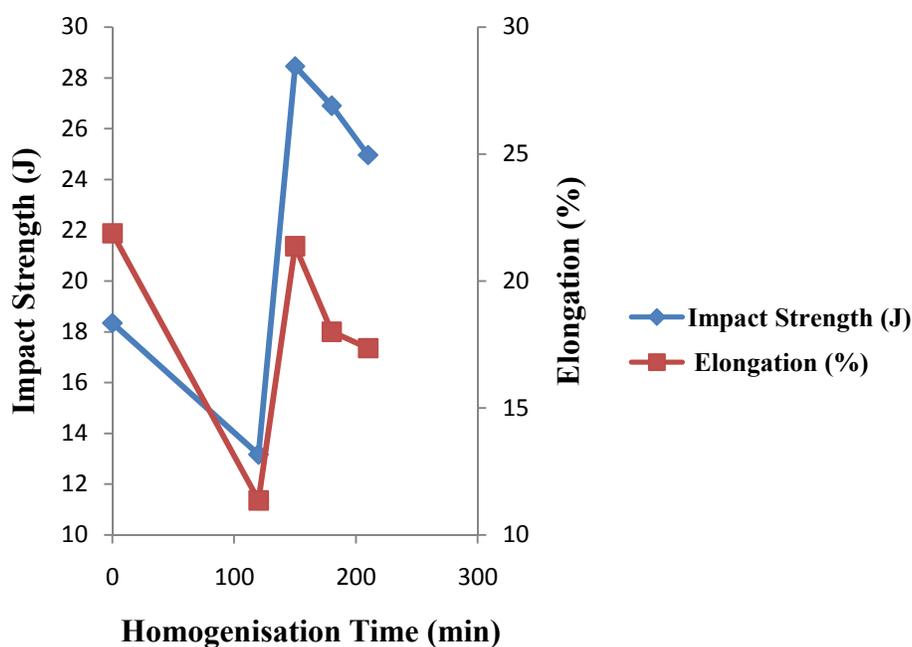
137 **Fig. 6. SEM micrographs of 6063 aluminium alloy homogenised at 580 °C for 3.5 hours and T6 tempered.**

138 The same result was observed for the hardness values as in the yield and ultimate tensile  
139 strengths in Fig. 1. Hardness values increased from 65 VHN (for no homogenisation) to 117  
140 VHN (for 2.5 hours homogenisation). Thereafter, it significantly dropped to 74 VHN and 69  
141 VHN for 3 and 3.5 hours homogenisation respectively. This reduction is likely to be due to the  
142 same reasons given above for drop in yield and ultimate tensile strengths. The high hardness  
143 value of samples homogenised for 2.5 hours prior to aging at 180 °C for 5 hours affirmed the  
144 theory that strength and hardness are constant multiples of each other [19, 20].

145 From Fig. 7, the ductility as indicated by the % elongation increased with homogenisation time.  
146 Generally, the % elongation was not significantly increased with homogenisation period and  
147 seemed to reach a low maximum level after 2.5 hours homogenisation. The comparatively high  
148 % elongation obtained when no prior homogenisation was carried out is not expected. It appears

149 that solution treatment alone resulted in stress relief annealing of the as-received structure Fig. 2.  
150 The % elongations values for specimens homogenised for 3 and 3.5 hours were still considerably  
151 higher than when no homogenisation was carried out. This was also confirmed by the  
152 microstructure in Fig. 3, where there is substantial quantity of unmodified second phase particles.  
153 Fig. 7 shows that, the impact strength increased with homogenisation time except for those  
154 samples homogenised for 2 hours and above reaching a maximum level for 2.5 hours. The low  
155 toughness values for samples homogenised for less than this period could be attributed to the  
156 presence of brittle intermetallic compounds present in their structures (Figures 2 and 3).

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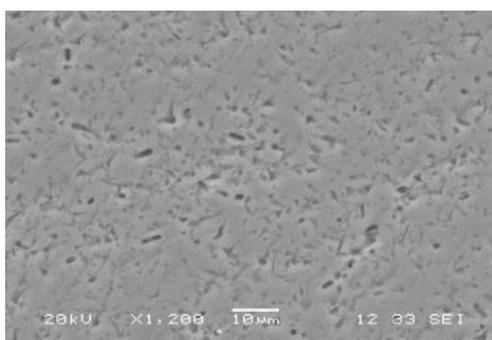
159 **Fig. 7. Influence of prior homogenisation treatment on impact strength and percentage elongation of T6**  
160 **tempered 6063 aluminium alloy.**

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162 The high toughness values observed for those samples homogenised for 2.5 hours and longer  
163 prior to solution treatment, were due to the high strength and high % elongation values obtained  
164 earlier. The comparatively low impact value for precipitation hardening without a prior  
165 homogenisation was as a result of very low tensile strength level caused by low homogenisation  
166 during solution treatment [5, 21].

167 The high toughness values observed for those samples homogenised for 2.5, 3 and 3.5 hours  
168 prior to aging were due to complete transformation of monoclinic  $\beta$ -Al-Fe-Si to a face-centered  
169 cubic  $\alpha$ -Al-Fe-Si phase and complete spheroidisation of other dispersoids present in this alloy  
170 (Figures 4, 5 and 6). The monoclinic  $\beta$ -Al-Fe-Si has low high-temperature ductility [16, 22].

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172

173 **Fig. 8. SEM micrograph of 6063 aluminium alloy homogenised at 580 °C for 2 hours and T6 tempered.**

174 The presence of manganese and chromium in this alloy also has a beneficial effect on its  
175 toughness. The presence of dispersoids in manganese/chromium containing alloys promotes  
176 intragranular precipitation and avoids precipitation on grain boundaries and formation of  
177 precipitate free zones adjacent to grain boundaries. This prevents weakening of grain boundaries  
178 and maintains the toughness of 6063 aluminium alloys [21]. This prevents the usual  
179 corresponding decrease in toughness and ductility of T6 tempered aluminium alloys. Because

180 those samples homogenised prior to aging have the highest number of grain boundaries,  
181 dislocation movement becomes more and more difficult during plastic deformation.

182 The strong increase in the mechanical properties as a result of homogenisation prior to aging can  
183 be attributed to the attainment of full potentials of homogenisation treatment. These according to  
184 the literatures [5,7] includes removal of microstructure inhomogeneities such as  
185 microsegregation, complete spheroidisation of the plate-like/sharp edges phases/dispersoids  
186 present in the as-received samples, formation of secondary dispersoids of favourable morphology  
187 and uniform distribution of alloying elements. Homogenisation ensures complete dissolution of  
188 sharp edge phases that are associated with direct chill casting of the ingot used for the extrusion  
189 of this alloys and formation of better ones at homogenisation temperature. All the alloying  
190 elements except for copper increase the quench sensitivity of this alloy [4]. These Al-Fe-Mn/Cr-  
191 Si dispersoids act as heterogeneous nucleants for magnesium-silicide precipitates during aging  
192 thereby increasing the quench sensitivity of this alloy. This leads to an increased in the density of  
193 precipitate formed. The finer these grains are the more the grain boundaries. During plastic  
194 deformation, dislocation movement must take place across these grain boundaries. Since  
195 polycrystalline grains are of different crystallographic orientations at the grain boundaries, a  
196 dislocation passing from one grain to another will have to change its direction of motion. Such  
197 changes of direction causes impediment to dislocation movement and thereby strengthening the  
198 alloy. Age hardening samples have the highest number of grain boundaries; dislocation  
199 movement becomes more and more difficult during plastic deformation. This is responsible for  
200 optimum combination of mechanical properties for those samples homogenised prior to aging as  
201 compared to those that were just T6 tempered.

202 The discrepancies in the yield strength of T6 tempered 6063 Al alloy (208 MPa) as compared to  
203 (215 MPa) in the standard data can be attributed to the presence of defects in the as-received  
204 alloy (Figure 2) which could not be completely eliminated by solution treatment alone as evident  
205 in the Figure 3 where there are still some unmodified plate-like phases. Also, there used to be  
206 discrepancies in the theoretical and ideal strengths of materials due to defects that are inherently  
207 associated with their production processes [23].

208 The EDX analysis of the phases present is presented in Figures 9 a, b and c respectively. Besides  
209 the prominent  $Mg_2Si$  phase, which serves as second phase particle and contributes to the final  
210 mechanical properties, Fe in combination with Si can formed the ternary phase  $AlFeSi$ , or else  
211 with Cr, Mn the quartenary phase  $AlFeCuMn$ . It also indicates the presence of  $CuAl_2$ . This is in  
212 agreement with Samaras and Haidemenopoulos [16] who stated that several phases may be  
213 present in commercial alloys containing Al–Mg–Si–Fe–Mn–Cu–Cr–Zn.

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(a)

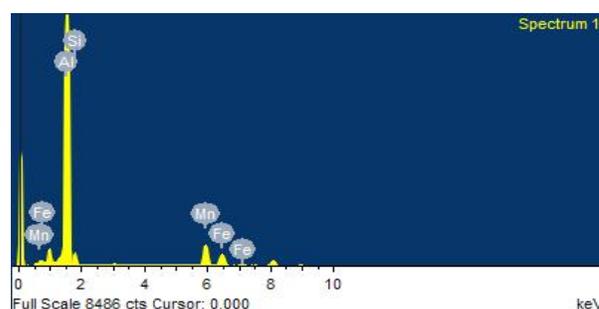
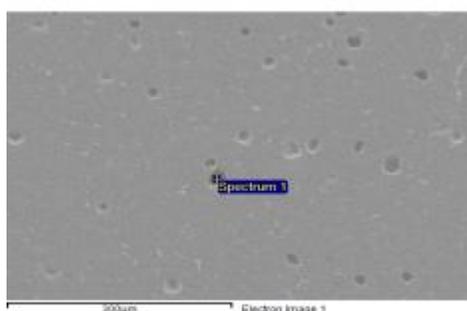
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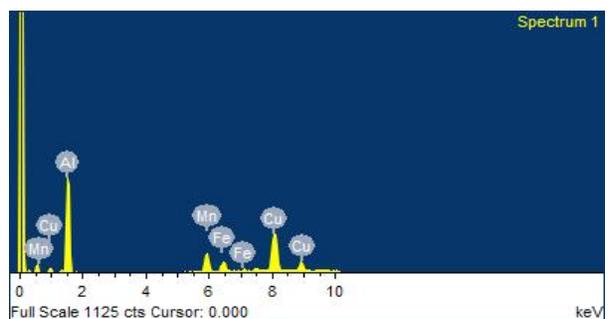
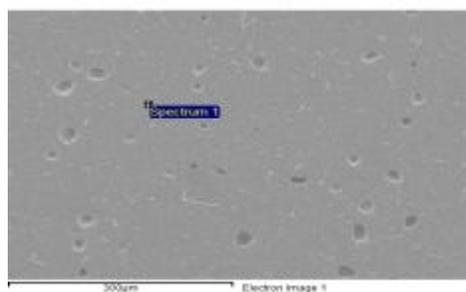
(b)

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(c)

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#### 4.0 CONCLUSION

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The following conclusions can be drawn from the results of this work.

241

1) The rod-like phase in the as-received 6063 Al alloy can be transformed into spherical phase by appropriate homogenisation treatment.

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243

2) Fragmentation and spheroidisation of sharp edges dispersoids in 6063 Al occur during homogenisation and greater degree of spheroidisation was achieved at 2.5, 3 and 3.5 hours of homogenisation at 580 °C.

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3) The optimum combination of strength and toughness values was obtained at 2.5 hours homogenisation prior to aging.

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4) Unlike in conventional aging where an increase in hardness and strength usually leads to corresponding decrease in ductility and toughness, homogenisation treatment prior to aging helps to maintain the ductility and toughness values of the age-hardened 6063 aluminium alloy.

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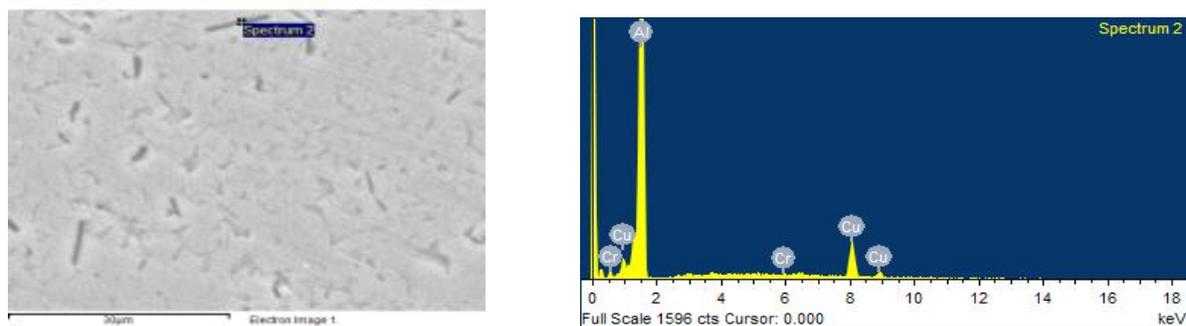


Fig. 9. a and b) shows the EDX analysis of spherical precipitates, c) shows the EDX analysis of Rod-like precipitates

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