# Original Research Article

# DYNAMIC BUCKLING LOAD OF AN IMPERFECT VISCOUSLY DAMPED SPHERICAL CAP STRESSED BY A STEP LOAD

## **ABSTRACT**

This paper determines the dynamic buckling load of a lightly and viscously damped imperfect spherical cap with a step load. The spherical cap is discretized into a pre-buckling symmetric mode and a buckling mode that consists of axisymmetric and non-axisymmetric buckling modes. The imperfection is taken at the shape of the buckling mode. The inherent problem contains a small parameter which necessitated the adoption of regular perturbation procedures, using asymptotic technique. The general result is designed to display the contributions of each of the terms in the governing differential equations. We deduce the results for the respective special cases where the axisymmetric imperfection parameter, namely  $\bar{\xi}_1$ , and the non-axisymmetric imperfection parameter  $\bar{\xi}_2$ , are zeros. We also determine the effects of each of the non-linear terms as well as the effects of the coupling term.

Keywords: Spherical cap, step load, dynamic buckling, imperfection parameter.

#### 1. INTRODUCTION.

The subject of dynamic buckling of elastic structures has been a thriving area of investigation ever since [1-3] developed the discipline of dynamic stability of elastic structures from the original static consideration that was prevalent before this time. Over the years, many investigations on dynamic stability of elastic structures have been added to the original sketchy and scattered pieces that saw the genesis of dynamic buckling of elastic structures as a research interest. Among the many scholarly investigations that have come to light include [4], [5], [6], [7] and [8], who investigated the dynamic buckling, of two-degree – of freedom systems with mode interaction under step loading. Mention must also be made of relatively recent investigations which include [9], who investigated the dynamic buckling of thin cylindrical shells under axial impact, [10-11], who studied the nonlinear dynamic buckling of stiffened plates under in-plane impact load.

But by far, the investigation that concerns us in this study is that by [12], who investigated the dynamic buckling loads of imperfection-sensitive structures from perturbation

procedures. His analysis was predicated primarily on the studies earlier enunciated by [1-3]. Other Pertinent investigations include those by [13], [14] and [15, 16], among others.

However, a cursory appraisal of all the investigations to date reveals that the phenomenon of damping has been given very little or no attention at all in the dynamic buckling process. We are of the strong opinion that since dynamic buckling process is a time dependent process, the effect of damping, no matter how slight, should not be overlooked. In this investigation, the presence of a small viscous damping is therefore assumed and given some level of prominence. Of course, the result obtained is far more representative of the actual physical life situation. To this end, we remark that a few of the many existing investigations that have tended to incorporate damping include the studies by [17-20], among others.

The layout of this investigation is as follows:

We shall first write down the mathematical equations satisfied by the structure investigated.

We shall next develop asymptotic techniques, using perturbation procedures to solve the governing equations analytically. We note that dynamic bucking problems are always non linear and therefore, closed-form exact solutions are not always possible. Therefore, regular perturbation method provides a suitable alternative to the solution of such problems, particularly when the problems contain small parameters in which asymptotic series expansion can always be invoked.

We shall lastly make pertinent deductions.

- 50 There are five sections in this paper. Section two examines the dynamic buckling load of an
- 51 imperfect viscously damped spherical cap stressed by a step load. Section three introduces
- 52 the viscous damping to Danielson's results. Section four considers the analysis of results
- while section five ends this work with a conclusion.

54 55

#### 2. THE DYNAMIC BUCKLING LOAD.

- Danielson, had, for simplicity, assumed that the normal displacement W(x, y, T) of the
- 58 spherical cap was given as

59 
$$W(x, y, T) = \xi_0(T)W_0(x, y) + \xi_1(T)W_1(x, y) + \xi_2(T)W_2(x, y)$$
 (1)

- where  $W_0(x, y)$  is the pre-buckling mode and  $W_1(x, y), W_2(x, y)$  are the axisymmetric and
- 61 non-axisymmetiic modes respectively.  $\xi_0(T), \xi_1(T)$  and  $\xi_2(T)$  are the respective time
- 62 dependent amplitudes of the associated modes. Imperfection W was introduced as

63 
$$\overline{W} = \overline{\xi}_1 W_1 + \overline{\xi}_2 W_2$$
 (2)

- where  $W_1,W_2$  still have meanings as before and  $ar{\xi}_1,ar{\xi}_2$  are the imperfect amplitudes
- assumed to be small relative to unity. On assuming suitable forms for  $W_0, W_1, W_2$  and
- substituting same into the compatibility and dynamic equilibrium equations and simplifying,
- 67 using his assumptions, Danielson obtained the following coupled differential equations for
- 68 step loading

69 
$$\frac{1}{\omega_0^2} \frac{d^2 \xi_0}{dT^2} + \xi_0 = \lambda f(T)$$
 (3)

70 
$$\frac{1}{\omega_1^2} \frac{d^2 \xi_1}{dT^2} + \xi_1 (1 - \xi_0) - k_1 \xi_1^2 + k_2 \xi_2^2 = \bar{\xi}_1 \xi_0$$
 (4)

71 
$$\frac{1}{\omega_2^2} \frac{d^2 \xi_2}{dT^2} + \xi_2 (1 - \xi_0) + \xi_1 \xi_2 = \bar{\xi}_2 \xi_0$$
 (5)

- 72  $\xi_i(0) = \xi_i'(0) = 0; i = 1, 2.$
- 73 Here, f(T) is the loading history which in our investigation, (as in Danielson's case), is the
- 74 step load characterized by

75 
$$f(T) = \begin{cases} 1, T > 0 \\ 0, T < 0 \end{cases}$$
, (6)

and,  $\lambda$  ,is the load parameter, considered to be non-dimensionalized and satisfies the

- 77 inequality  $0 < \lambda < 1$ .
- 78 In our guest for solution, we are to determine a particular value of  $\lambda$ , called the dynamic
- buckling load represented by  $\lambda_D$  and which satisfies the inequality  $0 < \lambda_D < 1$ . We define
- 80 the dynamic buckling load  $\lambda_D$  as the largest load parameter such that the solution to the
- damped version of problems (3)-(6) remains bounded for all time T>0. As in (3)-(5), we note
- that  $\omega_i; i=0,1,2$  are the circular frequencies of the associated modes  $\xi_0,\xi_1$  and  $\xi_2$
- 83 respectively while  $k_2$  and  $k_2$  are constants considered positive

84 85

#### 3. THE USE OF VISCOUS DAMPING IN DANIELSON'S RESULTS.

- 87 The present study is an extension of Danielson's problem to the case where a small viscous
- 88 damping is present. We however avoid Danielson's method (who used Mathieu type of
- 89 instability), for, as noted by [3, page 100], Mathieu type of instability is always associated
- 90 with many cycles of oscillations as opposed to just one shot of oscillation that triggers off
- 91 dynamic buckling.
- 92 For simplicity of analysis, we assume the existence of damping on the buckling modes.
- Since this damping must be only proportional to the velocity, we add the terms  $c_1 \frac{d\xi_1}{dT}$  and
- 94  $c_2 \frac{d\xi_2}{dT}$  to (4) and (5) respectively and the formulation now becomes

95 
$$\frac{1}{\omega_0^2} \frac{d^2 \xi_0}{dT^2} + \xi_0 = \lambda f(T)$$
 (7)

96 
$$\frac{1}{\omega_1^2} \frac{d^2 \xi_1}{dT^2} + c_1 \frac{d \xi_1}{dT} + \xi_1 (1 - \xi_0) - k_1 \xi_1^2 + k_2 \xi_2^2 = \bar{\xi}_1 \xi_0$$
 (8)

97 
$$\frac{1}{\omega_2^2} \frac{d^2 \xi_2}{dT^2} + c_2 \frac{d \xi_2}{dT} + \xi_2 (1 - \xi_0) + \xi_1 \xi_2 = \bar{\xi}_2 \xi_0$$
 (9)

- 98 where  $c_i$ , i = 1, 2 are the damping constants and which satisfy the inequality  $0 < c_i < 1$ .
- 99 Using f(T)=1 and substituting (6) into (7) we have

100 
$$\frac{1}{\omega_0^2} \frac{d^2 \xi_0}{dT^2} + \xi_0 = \lambda$$
 (10)

101 
$$\frac{1}{\omega^2} \frac{d^2 \xi_1}{dT^2} + c_1 \frac{d\xi_1}{dT} + \xi_1 (1 - \xi_0) - k_1 \xi_1^2 + k_2 \xi_2^2 = \bar{\xi}_1 \xi_0$$
 (11)

102 
$$\frac{1}{\omega_1^2} \frac{d^2 \xi_2}{dT^2} + c_2 \frac{d \xi_2}{dT} + \xi_2 (1 - \xi_0) + \xi_1 \xi_2 = \bar{\xi}_2 \xi_0$$
 (12)

103 Now using,

104 
$$t = \omega_0 T$$
,

105 so that

106 
$$\frac{d()}{dT} = \omega_0 \frac{d()}{dt}, \frac{d^2()}{dT^2} = \omega_0^2 \frac{d^2()}{dt^2},$$

107 Then (10)-(12) become

108 
$$\frac{d^2 \xi_0}{dt^2} + \xi_0 = \lambda \tag{13}$$

109

$$110 \qquad \frac{d^2 \xi_1}{dt^2} + \left[ \frac{c_1 \omega_0 \omega_1^2}{\omega_0^2} \right] \frac{d \xi_1}{dt} + \left[ \frac{\omega_1}{\omega_0} \right]^2 \xi_1 (1 - \xi_0) - \left[ \frac{\omega_1}{\omega_0} \right]^2 k_1 \xi_1^2 + \left[ \frac{\omega_1}{\omega_0} \right]^2 k_2 \xi_2^2 = \left[ \frac{\omega_1}{\omega_0} \right]^2 \bar{\xi}_1 \xi_0 \qquad (14)$$

111 
$$\frac{d^2 \xi_2}{dt^2} + \left[ \frac{c_2 \omega_0 \omega_2^2}{\omega_0^2} \right] \frac{d \xi_2}{dt} + \left[ \frac{\omega_2}{\omega_0} \right]^2 \xi_2 (1 - \xi_0) + \left[ \frac{\omega_2}{\omega_0} \right]^2 \xi_1 \xi_2 = \left[ \frac{\omega_2}{\omega_0} \right]^2 \bar{\xi}_2 \xi_0$$
 (15)

112 Next, we let

113 
$$2\alpha_1 \varepsilon = \frac{c_1 \omega_1^2}{\omega_0}, 2\alpha_2 \varepsilon = \frac{c_2 \omega_2^2}{\omega_0}, Q = \frac{\omega_1}{\omega_0}, R = \frac{\omega_2}{\omega_0}, S = \left[\frac{\omega_2}{\omega_1}\right]^2$$
 (16)

114 where

115 
$$\varepsilon = \lambda Q^2 = \lambda \left[ \frac{\omega_1}{\omega_0} \right]^2, \tag{17}$$

116 and

117 
$$0 < \alpha_1 < 1, \ 0 < \alpha_2 < 1, \ 0 < Q < 1, \ 0 < R < 1$$
 and  $0 < \varepsilon < 1$ 

118 Substituting (16) into (14) and (15) yield

119 
$$\frac{d^2 \xi_0}{dt^2} + \xi_0 = \lambda \tag{18}$$

120 
$$\frac{d^2 \xi_1}{dt^2} + 2\alpha_1 \varepsilon \frac{d\xi_1}{dt} + Q^2 \xi_1 (1 - \xi_0) - k_1 Q^2 \xi_1^2 + k_2 Q^2 \xi_2^2 = Q^2 \bar{\xi}_1 \xi_0$$
 (19)

121 
$$\frac{d^2 \xi_2}{dt^2} + 2\alpha_2 \varepsilon \frac{d\xi_2}{dt} + R^2 \xi_2 (1 - \xi_0) + R^2 \xi_1 \xi_2 = R^2 \bar{\xi}_2 \xi_0$$
 (20)

122 
$$\xi_i(0) = \xi_i'(0) = o; i = 1, 2.$$

123 As in  $\left[1-3\right]$ , we neglect the pre-buckling inertia term, so that from (18) we get

$$124 \xi_0 = \lambda (21)$$

- Here, we assume zero pre-buckling inertia term since the load imparts zero initial
- displacement and velocity to the pre-buckling mode
- 127 On simplification, using (21), equations (19) and (20) yield

128 
$$\frac{d^2 \xi_1}{dt^2} + 2\alpha_1 \varepsilon \frac{d\xi_1}{dt} + Q^2 \xi_1 - \varepsilon \xi_1 - k_1 Q^2 \xi_1^2 + k_2 Q^2 \xi_2^2 = \varepsilon \bar{\xi}_1$$
 (22)

129 and

$$130 \qquad \frac{d^2 \xi_2}{dt^2} + 2\alpha_2 \varepsilon \frac{d\xi_2}{dt} + R^2 \xi_2 - \varepsilon S \xi_2 + R^2 \xi_1 \xi_2 = \varepsilon S \bar{\xi}_2$$
 (23)

131 
$$\xi_i(0) = \xi_i'(0) = 0; i = 1, 2$$

where.

133 
$$S = \left\lceil \frac{R}{Q} \right\rceil^2$$
.

134 We assume a small time scale  $\tau$  such that,

135 
$$\tau = \mathcal{E}t$$
 (24a)

136 and

137 
$$\xi_i' = \xi_{i,t} + \mathcal{E}\xi_{i,\tau} \tag{24b}$$

138 
$$\xi_i^{"} = \xi_{i,tt} + 2\varepsilon \xi_{i,t\tau} + \varepsilon^2 \xi_{i,\tau\tau}; i = 1,2$$
 (24c)

139 We denote our perturbation parameter by  $\mathcal{E}$  so that

140 
$$\xi_1(t) = \sum_{i=1}^{\infty} \zeta^{(i)}(t, \tau) \varepsilon^i$$
 (25)

141 
$$\xi_2(t) = \sum_{i=1}^{\infty} \eta^{(i)}(t, \tau) \varepsilon^i$$
 (26)

Substituting (25) and (26) into (22) and (23), using (24b) and (24c), equating terms of the

143 orders of  $\mathcal{E}$  we get,

144 
$$\varsigma_{.tt}^{(1)} + Q^2 \varsigma^{(1)} = \bar{\xi}_1$$
 (27)

145 
$$\varsigma_{,tt}^{(2)} + Q^2 \varsigma^{(2)} = -2\alpha_1 \varsigma_{,t}^{(1)} + \varsigma^{(1)} + k_1 Q^2 \varsigma^{(1)^2} - k_2 Q^2 \eta^{(1)^2} - 2\varsigma_{,t\tau}^{(1)}$$
 (28)

146 and

147 
$$\eta_{tt}^{(1)} + R^2 \eta^{(1)} = S \bar{\xi}_2$$
 (29)

148 
$$\eta_{,tt}^{(2)} + R^2 \eta^{(2)} = -2\alpha_2 \eta_{,t}^{(1)} + S \eta^{(1)} - 2\eta_{,t\tau}^{(1)} - R^2 \varsigma^{(1)} \eta^{(1)}$$
 (30)

149 
$$\varsigma^{(i)}(0,0) = \eta^{(i)}(0,0) = 0, i = 1,2$$
 (31)

150 
$$\varsigma_{,t}^{(i)}(0,0) = \eta_{,t}^{(i)}(0,0) = 0, i = 1,2$$
 (32)

151 
$$\varsigma_{,t}^{(i+1)}(0,0) + \varsigma_{,\tau}^{(i)}(0,0) = \eta_{,t}^{(i+1)}(0,0) + \eta_{,\tau}^{(i)}(0,0) = 0, i = 1,2$$
 (33)

152 The solution of (27) using (31) and (32) is

153 
$$\varsigma^{(1)}(t,\tau) = a_1(\tau)\cos Qt + b_1(\tau)\sin Qt + \frac{\bar{\xi}_1}{Q^2}$$
 (34a)

154 
$$a_1(0) = -\frac{\bar{\xi}_1}{Q^2}; b_1(0) = 0$$
 (34b)

155 Similarly, the solution of (28) is

156 
$$\eta^{(1)}(t,\tau) = a_2(\tau)\cos Rt + b_2(\tau)\sin Rt + \frac{S\xi_2}{R^2}$$
 (35a)

157 
$$a_2(0) = -\frac{S\bar{\xi}_2}{R^2}; b_2(0) = 0$$
 (35b)

158 Substituting using (34a) and (35a) into (28), we have

159 
$$\varsigma_{,tt}^{(2)} + Q^2 \varsigma^{(2)} = -2\alpha_1 \left[ -Qa_1 \sin Qt + Qb_1 \cos Qt \right] + \left[ a_1 \cos Qt + b_1 \sin Qt + \frac{\bar{\xi}_1}{Q^2} \right]$$

$$-k_2 Q^2 \left[ \frac{1}{2} [a_2^2 + b_2^2] + a_2 b_2 \sin 2Rt + \frac{1}{2} [a_2^2 - b_2^2] \cos 2Rt \right]$$

$$-k_{2}Q^{2} \left[ + \frac{2a_{2}S\bar{\xi}_{2}}{R^{2}}\cos Rt + \frac{2b_{2}S\bar{\xi}_{2}}{R^{2}}\sin Rt \right]$$

$$+k_{1}Q^{2}\left[\frac{1}{2}\left[a_{1}^{2}+b_{1}^{2}\right]+a_{1}b_{1}\sin 2Qt+\frac{1}{2}\left[a_{1}^{2}-b_{1}^{2}\right]\cos 2Qt\right]$$

163 
$$+k_1 Q^2 \left[ +\frac{2a_1\bar{\xi}_1}{Q^2}\cos Qt + \frac{2b_1\bar{\xi}_1}{Q^2}\sin Qt \right] + 2Q\left[a_1\sin Qt - b_1\cos Qt\right]$$
 (36)

- To maintain a uniformly valid asymptotic solution in time scale t, we equate the coefficients
- of  $\cos Qt$  and  $\sin Qt$  to zero to get (on the rhs of 36). This ensures a finite at infinite time,
- i.e. as t tends to infinity, such terms tends to zero. The terms are called secular terms, we
- therefore device a way of eliminating the terms, hence making the solution bounded for all *t*

168 
$$\cos Qt :- 2\alpha_1 b_1 Q + a_1 + \frac{2a_1 k_1 \bar{\xi}_1 Q^2}{Q^2} - 2b_1 Q = 0$$

169 On simplification, we get

170 
$$b_1' + \alpha_1 b_1 = \frac{a_1}{2Q} \left[ 1 + 2k_1 \, \bar{\xi}_1 \right]$$

171 
$$b_1' + \alpha_1 b_1 = a_1 \varphi$$
 (37a)

172 and similarly,

173 
$$\sin Qt :- 2\alpha_1 a_1 Q + b_1 + \frac{2b_1 k_1 \bar{\xi}_1 Q^2}{Q^2} + 2a_1 Q = 0$$

174 Simplifying gives

175 
$$a_1' + \alpha_1 a_1 = -\frac{b_1}{2Q} \left[ 1 + 2k_1 \bar{\xi}_1 \right]$$

176 
$$a_1' + \alpha_1 a_1 = -b_1 \varphi$$
 (37b)

177 where,

$$178 \qquad \left( \begin{array}{c} \dot{} = \frac{d()}{d\tau}, \end{array} \right)$$

179 and

180 
$$\varphi = \frac{1}{2Q} \left[ 1 + 2k_1 \, \bar{\xi}_1 \right]$$

181 Simplification of (37a, b) yield

182 
$$b_1^{\prime\prime} + \alpha_1 b_1^{\prime} = -\varphi \left[ b_1 \left[ \varphi + \frac{\alpha_1^2}{\varphi} \right] + \frac{\alpha_1 b_1^{\prime}}{\varphi} \right]$$

183 
$$b_1^{"} + 2\alpha_1 b_1^{"} + \varphi b_1 \left[ \varphi + \frac{\alpha_1^2}{\varphi} \right] = 0$$

184 
$$b_1(0) = 0; b_1'(0) = -\frac{\varphi \bar{\xi}_1}{O^2}$$
 (37c)

185 and

186 
$$a_1'' + \alpha_1 a_1' = -\varphi \left[ a_1 \left[ \varphi + \frac{\alpha_1^2}{\varphi} \right] + \frac{\alpha_1 a_1'}{\varphi} \right]$$

187 
$$a_1'' + 2\alpha_1 a_1' + \varphi a_1 \left[ \varphi + \frac{\alpha_1^2}{\varphi} \right] = 0$$

188 
$$a_1(0) = -\frac{\xi_1}{Q^2}; a_1'(0) = \frac{\alpha_1 \xi_1}{Q^2}$$
 (37d)

The remaining part of the equation in the substitution into (28) as obtained from (36) is

190 
$$\varsigma_{.tt}^{(2)} + Q^2 \varsigma^{(2)} = q_1 + k_1 Q^2 [p_0(\tau) \sin 2Qt + p_1(\tau) \cos 2Qt]$$

191 
$$-k_2Q^2[p_2(\tau)\sin 2Rt + p_3(\tau)\cos 2Rt + p_4(\tau)\cos Rt + p_5(\tau)\sin Rt]$$
 (38a)

192 
$$\zeta^{(2)}(0,0) = 0; \zeta_t^{(2)}(0,0) + \zeta_\tau^{(2)}(0,0) = 0$$
 (38b)

193 where,

194 
$$q_1 = \frac{\bar{\xi}_1}{Q^2} + k_1 Q^2 r_0(\tau) - k_2 Q^2 r_1(\tau); p_0(\tau) = a_1 b_1; p_1(\tau) = \frac{1}{2} \left[ a_1^2 - b_1^2 \right]$$
 (38c)

195 
$$p_2(\tau) = a_2 b_2; p_3(\tau) = \frac{1}{2} \left[ a_2^2 - b_2^2 \right]; p_4(\tau) = \frac{2a_2 S \xi_2}{R^2}; p_5(\tau) = \frac{2b_2 S \xi_2}{R^2}$$
 (38d)

196 
$$r_0(\tau) = \frac{1}{2} \left[ a_2^2 + b_2^2 \right] r_1(\tau) = \frac{1}{2} \left[ a_1^2 + b_1^2 \right]$$
 (38e)

197 
$$p_0(0) = 0; p_1(0) = \frac{\bar{\xi}_1^2}{2Q^4}; p_2(0) = 0; p_3(0) = \frac{S^2 \bar{\xi}_2^2}{2R^4}$$
 (38f)

198 
$$p_4(0) = \frac{2S^2 \bar{\xi}_2^2}{R^4}; p_5(0) = 0; r_0(0) = \frac{\bar{\xi}_1^2}{2Q^4}; r_1(0) = \frac{S^2 \bar{\xi}_2^2}{2R^4};$$
 (38g)

199 The solution of (38a), using (38b) is

200 
$$\varsigma^{(2)}(t,\tau) = a_3(\tau)\cos Qt + b_3(\tau)\sin Qt + \frac{q_1}{Q^2} - \frac{k_1}{3}[p_6(\tau)\sin 2Qt + p_7(\tau)\cos 2Qt]$$

201

$$-k_2 Q^2 \left[ p_8(\tau) \sin 2Rt + p_9(\tau) \cos 2Rt + p_{10}(\tau) \cos Rt + p_{11}(\tau) \sin Rt \right]$$
 (39a)

203 
$$a_3(0) = \bar{\xi}_1 l_0 + k_1 \bar{\xi}_1^2 l_1 + k_2 \bar{\xi}_2 \left[ \frac{S}{R^2} \right]^2 l_2; b_3(0) = -\frac{\alpha_1 \bar{\xi}_1}{Q^3}$$
 (39b)

204 where

$$205 l_0 = -\frac{1}{Q^4}; l_1 = -\frac{1}{2Q^2} + \frac{1}{6Q^4}; l_2 = \frac{1}{2} + \frac{1}{2[Q^2 - 4R^2]} - \frac{2}{R^2[Q^2 - 4R^2]}$$
 (39c)

206 
$$p_6(\tau) = a_1 b_1; p_7(\tau) = a_2 b_2; p_8(\tau) = \frac{p_2(\tau)}{Q^2 - 4R^2}; p_9(\tau) = \frac{p_3(\tau)}{Q^2 - 4R^2}$$
 (39d)

207 
$$p_{10}(\tau) = \frac{p_4(\tau)}{Q^2 - R^2}; p_{11}(\tau) = \frac{p_5(\tau)}{Q^2 - R^2}; p_{11}(0) = 0$$
 (39e)

208 
$$p_6(0) = 0; p_7(0) = \frac{\bar{\xi}_1^2}{2Q^4}; p_8(0) = 0; p_9(0) = \frac{S^2 \xi_2^2}{2R^4 [Q^2 - 4R^2]}; p_{10}(0) = \frac{2S^2 \bar{\xi}_2^2}{R^4 [Q^2 - R^2]}$$
 (39f)

$$\eta_{tt}^{(2)} + R^2 \eta^{(2)} = -2\alpha_2 [-Ra_2 \sin Rt + Rb_2 \cos Rt] - 2R[-a_2 \sin Rt + b_2 \cos Rt] + S$$

$$210 \qquad \left[ a_2 \cos Rt + b_2 \sin Rt + \frac{S\xi_2}{R^2} \right] - \frac{R^2}{2}$$

$$+[a_1a_2-b_1b_2]\cos[Q-R]t+[a_1b_2+b_1a_2]\sin[Q+R]t$$

$$+[a_1a_2+b_1b_2]\cos[Q-R]t+[b_1a_2-a_1b_2]\sin[Q-R]t$$

- Now, to ensure a uniformly valid asymptotic solution in tine scale t, we equate the
- coefficients of  $\cos Rt$  and  $\sin Rt$  to zero. This will ensure a finite at infinite time, i.e. as t

(40)

- tends to infinity, such terms tends to zero thereby making the solution not to be bounded,
- 215 hence non-uniform. Such terms are called secular terms and our aim is to get rid of them.

216 
$$\cos Rt :- 2\alpha_2 b_2 R - 2b_2' R + S\alpha_2 - \frac{a_2 \bar{\xi}_1 R^2}{Q^2} = 0$$

217 Implies that,

218 
$$b_2' + \alpha_2 b_2 = \frac{a_2}{2R} \left[ S - \frac{\bar{\xi}_1 R^2}{Q^2} \right]$$

219 
$$b_2' + \alpha_2 b_2 = a_2 \Phi$$
 (41a)

220 and similarly,

221 
$$\sin Rt := 2\alpha_2 a_2 R + 2a_2' R + Sb_2 - \frac{b_2 \bar{\xi}_1 R^2}{Q^2} = 0$$

222 Simplifying gives

223 
$$a_2' + \alpha_2 a_2 = -\frac{b_2}{2R} \left[ S - \frac{\bar{\xi} R^2}{Q^2} \right]$$

225 
$$a_2' + \alpha_2 a_2 = -b_2 \Phi$$
 (41b)

226 where,

227 
$$\Phi = \frac{1}{2R} \left[ S - \frac{\bar{\xi}_1 R^2}{Q^2} \right].$$

228 Simplification of (41a, b) yield

229 
$$b_2'' + \alpha_2 b_2' = -\Phi[\Phi b_2 + \alpha_2 a_2]$$

230 
$$b_2^{"} + \alpha_2 b_2^{'} = -\Phi \left[ \Phi b_2 + \frac{\alpha_2}{\Phi} \left[ b_2^{'} + \alpha_2 b_2 \right] \right]$$

231 
$$b_2'' + 2\alpha_2 b_2' + b_2 [\Phi^2 + \alpha_2^2] = 0$$

232 
$$b_2(0) = 0; b_2'(0) = -\frac{\Phi S \xi_2}{R^2}$$
 (41c)

233 and

234 
$$a_2'' + \alpha_2 a_2' = -\Phi[\Phi a_2 - \alpha_2 b_2]$$

235 
$$a_2'' + \alpha_2 a_2' = -\Phi \left[ \Phi a_2 + \frac{\alpha_2}{\Phi} \left[ a_2' + \alpha_2 a_2 \right] \right]$$

236 
$$a_2'' + 2\alpha_2 a_2' + a_2 \left[ \Phi^2 + \alpha_2^2 \right] = 0$$

237 
$$a_2(0) = -\frac{S\bar{\xi}_2}{R^2}; a_2'(0) = -\frac{\alpha_2 S\bar{\xi}_2}{R^2}$$
 (41d)

238 The remaining part of the equation in the substitution into (30) as obtained from (40) is

240  $\eta_{,tt}^{(2)} + R^2 \eta^{(2)} = q_2 - \frac{R^2}{2} \begin{bmatrix} p_{12}(\tau)\cos Qt + p_{13}(\tau)\sin Qt + p_{14}(\tau)\cos[Q+R]t \\ + p_{15}(\tau)\sin[Q+R]t + p_{16}(\tau)\cos[Q-R]t + p_{17}(\tau)\sin[Q-R] \end{bmatrix}$ 

241 
$$\eta^{(2)}(0,0) = 0; \eta_{,t}^{(2)}(0,0) + \eta_{,\tau}^{(1)}(0,0) = 0$$
 (42b)

242 where,

243 
$$q_2 = \frac{S^2 \bar{\xi_2}}{R^2} - \frac{S \bar{\xi_1} \bar{\xi_2}}{O^2}; p_{12}(\tau) = \frac{2a_1 S \bar{\xi_2}}{R^2}; p_{13}(\tau) = \frac{2b_1 S \bar{\xi_2}}{R^2}; p_{14}(\tau) = a_1 a_2 - b_1 b_2$$
 (42c)

244

245 
$$p_{15}(\tau) = b_1 b_2 + b_1 a_2; p_{16}(\tau) = a_1 a_2 + b_1 b_2; p_{17}(\tau) = b_1 a_2 - a_1 b_1$$
 (42d)

246 
$$p_{12}(0) = \frac{2S\bar{\xi}_1\bar{\xi}_2}{Q^2R^2}; p_{13}(0) = 0; p_{14}(0) = \frac{S\bar{\xi}_1\bar{\xi}_2}{Q^2R^2}; p_{15}(0) = 0;$$
 (42e)

247 
$$p_{16}(0) = \frac{S\bar{\xi}_1\bar{\xi}_2}{O^2R^2}; p_{17}(0) = 0$$
 (42f)

248 The solution of (42a) using (42b) is

249 
$$\eta^{(2)}(t,\tau) = a_4(\tau)\cos Rt + b_4(\tau)\sin Rt + \frac{q_2}{R^2} - \frac{1}{2} \begin{bmatrix} p_{18}(\tau)\cos Qt + p_{19}(\tau)\sin Qt + \\ p_{20}(\tau)\cos[Q+R]t + p_{21}(\tau)\sin[Q+R]t + \\ p_{22}(\tau)\cos[Q-R]t + p_{23}(\tau)\sin[Q-R]t \end{bmatrix}$$
(43a)

250 
$$a_4(0) = S^2 \bar{\xi}_2 l_3 + R^2 S \bar{\xi}_1 \bar{\xi}_2 l_4; b_4(0) = -\frac{\alpha_2 S \xi_2}{R^3}$$
 (43b)

251 where,

$$252 l_4 = \left[ \frac{1}{\left[ R^2 Q \right]^2} + \frac{1}{2} \left[ \frac{-2}{\left[ R Q \right]^2 \left[ R^2 - Q^2 \right]} - \frac{1}{Q \left[ R Q \right]^2 \left[ 2R + Q \right]} + \frac{1}{Q \left[ R Q \right]^2 \left[ 2R - Q \right]} \right] \right] (43c)$$

253

254 
$$l_3 = -\frac{1}{R^4}; p_{18}(\tau) = \frac{p_{12}(\tau)}{R^2 - Q^2}; p_{19}(\tau) = \frac{p_{13}(\tau)}{R^2 - Q^2}; p_{20}(\tau) = \frac{p_{14}(\tau)}{Q[2R + Q]}$$
 (43d)

255

$$p_{21}(\tau) = \frac{p_{15}(\tau)}{Q[2R+Q]}; p_{22}(\tau) = \frac{p_{16}(\tau)}{Q[2R-Q]}; p_{23}(\tau) = \frac{p_{17}(\tau)}{Q[2R-Q]}$$
(43e)

257 
$$p_{18}(0) = \frac{2S\bar{\xi}_1\bar{\xi}_2}{Q^2R^2[R^2-Q^2]}; p_{19}(0) = 0; p_{20}(0) = \frac{S\bar{\xi}_1\bar{\xi}_2}{Q^3R^2[2R+Q]}$$
 (43f)

258

259 
$$p_{21}(0) = 0; p_{22}(0) = \frac{S\bar{\xi}_1\bar{\xi}_2}{Q^2R^2[2R-Q]}; p_{23}(0) = 0$$
 (43g)

260

Next, using (34a), (39a) and (35a), (43a) we deduce the displacements as

262 
$$\xi_1(t) = \zeta^{(1)}(t,\tau)\varepsilon + \zeta^{(2)}(t,\tau)\varepsilon^2 + \dots$$
 (44a)

263 and

264 
$$\xi_2(t) = \eta^{(1)}(t,\tau)\varepsilon + \eta^{(2)}(t,\tau)\varepsilon^2 + \dots$$
 (44b)

- We seek the maximum displacement for both  $\xi_1(t)$  and  $\xi_2(t)$ . To achieve this, we shall first
- determine the critical values of t and  $\tau$  for each of  $\xi_1(t)$  and  $\xi_2(t)$  at their maximum values.
- The condition for the maximum displacements of  $\xi_1(t)$  and  $\xi_2(t)$  is obtain from (24b).hence

$$\xi_{1,t} + \varepsilon \xi_{1,\tau}, \tag{45a}$$

$$\xi_{2,t} + \varepsilon \xi_{2,\tau}, \tag{45b}$$

We know from (44a, b) that

271 
$$\xi_1(t) = \zeta^{(1)}(t,\tau)\varepsilon + \zeta^{(2)}(t,\tau)\varepsilon^2 + \dots$$
 (46a)

272 
$$\xi_2(t) = \eta^{(1)}(t,\tau)\varepsilon + \eta^{(2)}(t,\tau)\varepsilon^2 + \dots$$
 (46b)

273 On applying (45a, b) to (46a, b), we get

274 
$$\varsigma_{,t} + \varepsilon \varsigma_{,\tau} = \left[ \varsigma_{,t}^{(1)}(t_a, \tau_a)\varepsilon + \varsigma_{,t}^{(2)}(t_a, \tau_a)\varepsilon^2 + \ldots \right]$$

275

$$+\varepsilon \left[\varsigma_{,\tau}^{(1)}(t_a,\tau_a)\varepsilon + \varsigma_{,\tau}^{(2)}(t_a,\tau_a)\varepsilon^2 + \ldots\right] = 0$$
(47a)

277 and

278 
$$\eta_{,t} + \varepsilon \eta_{,\tau} = \left[ \eta_{,t}^{(1)} (T_c, \tau_c) \varepsilon + \eta_{,t}^{(2)} (T_c, \tau_c) \varepsilon^2 + \ldots \right]$$

279

$$+\varepsilon \left[ \eta_{\tau}^{(1)}(T_c, \tau_c) \varepsilon + \eta_{\tau}^{(2)}(T_c, \tau_c) \varepsilon^2 + \dots \right] = 0$$

$$(47b)$$

- where,  $\left(t_a, \tau_a\right)$  and  $\left(T_c, \tau_c\right)$  are the values of t and  $\, au$  at the maximum displacement of
- 282  $\zeta(t,\tau)$  and  $\eta(t,\tau)$  respectively.
- We now expand (47a, b) in a Taylor series about  $t_a=t_0, au_a=0$  and  $T_c=T_0, au_c=0$  ,and
- 284 thereafter equate to zero the terms of the same orders of  $\,\mathcal{E}\,$  to get

285 
$$\varsigma_{,t}^{(1)}(t_0,0) = 0$$
 (48a)

286 
$$t_1 \varsigma_{,tt}^{(1)}(t_0,0) + t_0 \varsigma_{,tt}^{(1)}(t_0,0) + \varsigma_{,t}^{(2)}(t_0,0) + \varsigma_{,t}^{(1)}(t_0,0) = 0$$
 (48b)

287 and

288 
$$\eta_{,t}^{(1)}(T_0,0) = 0$$
 (49a)

289 
$$T_1 \eta_{,tt}^{(1)}(T_0,0) + T_0 \eta_{,t\tau}^{(1)}(T_0,0) + \eta_{,t}^{(2)}(T_0,0) + \eta_{,\tau}^{(1)}(T_0,0) = 0$$
 (49b)

290 Substituting for  $\varsigma_{t}^{(1)}$  from (34a) in (48a) and simplifying we get

$$\sin Qt_0 = 0 \tag{50a}$$

292 A further simplification of (50a) gives

$$t_0 = \frac{\pi}{Q} \tag{50b}$$

294 A similar solution for (49a) is

$$T_0 = \frac{\pi}{R}$$
 (50c)

296 Next, we deduce from (48b) that

297 
$$t_{1} = -\frac{1}{\varsigma_{,t}^{(1)}(t_{0},0)} \left[ t_{0} \varsigma_{,t\tau}^{(1)}(t_{0},0) + \varsigma_{,t}^{(2)}(t_{0},0) + \varsigma_{,\tau}^{(1)}(t_{0},0) \right]$$
 (51a)

298 Simplification of the following terms are however necessary in this analysis,

299 
$$\varsigma_{,t}^{(2)}(t_0,0) = \alpha_1 \bar{\xi}_1 l_5 + k_1 \bar{\xi}_1^2 l_6 - k_2 S^2 \bar{\xi}_2^2 l_7; \varsigma_{,t}^{(1)}(t_0,0) = \bar{\xi}_1 l_8$$
 (51b)

300 
$$\varsigma_{,\tau}^{(1)}(t_0,0) = \bar{\xi}_1 l_9; \varsigma_{,tt}^{(1)}(t_0,0) = \bar{\xi}_1; \varsigma^{(1)}(t_0,0) = 2\bar{\xi}_1 l_{10}$$
 (51c)

301 
$$\varsigma^{(2)}(t_0,0) = 2\bar{\xi}_1 l_{11} + k_1 \bar{\xi}_1^2 l_{12} + k_2 \bar{\xi}_2^2 \left[ \frac{S}{R^2} \right]^2 l_{13}; \varsigma_{,i}^{(1)}(t_0,0) = 0;$$
 (51d)

302 where

303 
$$l_5 = \frac{1}{Q^2}; l_6 = \frac{Sin2Qt_0}{3Q^2}; l_7 = \frac{Q^2Sin2Rt_0}{R^3[Q^2 - 4R^2]}$$
 (51e)

304

$$305 \qquad l_8 = \frac{\varphi}{Q}; l_9 = -\frac{\alpha_1}{Q^2}; l_{10} = \frac{1}{Q^2}; l_{11} = \frac{1}{Q^4}; l_{12} = \frac{1}{Q^2} - \frac{1}{3Q^4}$$
 (51f)

306

308

309 On substituting (51, b-d) on (51a), we have

310 
$$t_1 = \alpha_1 l_5 + k_1 \xi_1 l_6 - k_2 S \xi_2 l_7 + t_0 l_8 + l_9$$
 (52)

311 Similarly, deducing from (49b) yields

312 
$$T_{1} = -\frac{1}{\eta_{tt}^{(1)}(T_{0},0)} \left[ T_{0} \eta_{tt}^{(1)}(T_{0},0) + \eta_{tt}^{(2)}(T_{0},0) + \eta_{tt}^{(1)}(t_{0},0) \right]$$
 (53a)

313 We however note the following simplifications

314 
$$\eta_{t}^{(2)}(T_{0},0) = \alpha_{2}S\bar{\xi}_{2}l_{14} + S^{2}\bar{\xi}_{2}l_{15} + S\bar{\xi}_{1}\bar{\xi}_{2}l_{16}; \eta_{t\pi}^{(1)}(T_{0},0) = S\bar{\xi}_{2}l_{17}$$
 (53b)

315 
$$\eta_x^{(1)}(T_0,0) = S^2 \bar{\xi}_2 l_{18}; \eta_x^{(1)}(T_0,0) = -S \bar{\xi}_2; \eta^{(1)}(T_0,0) = 2S \bar{\xi}_2 l_{20}$$
 (53c)

316 
$$\eta^{(2)}(T_0,0) = S^2 \bar{\xi}_2 l_{21} + R^2 S \bar{\xi}_1 \bar{\xi}_2 l_{19}; \eta_{,t}^{(1)}(T_0,0) = 0;$$
 (53d)

317 where

318 
$$l_{14} = -\frac{\cos RT_0}{R^2}; l_{15} = -Rl_3 \sin RT_0$$
 (53e)

319 
$$l_{16} = -R^{3}Sl_{4}\sin RT_{0} - \frac{R^{2}}{2} \begin{vmatrix} \frac{2\sin QT_{0}}{QR^{2}[R^{2} - Q^{2}]} - \frac{\cos QT_{0}}{[RQ]^{2}[2R + Q]} \\ -\frac{[Q + R]\sin[Q + R]T_{0}}{Q[RQ]^{2}[2R + Q]} - \frac{[Q - R]\sin[Q - R]T_{0}}{Q[RQ]^{2}[2R - Q]} \end{vmatrix}$$
 (53f)

320 
$$l_{17} = \frac{\Phi}{R}; l_{18} = -\frac{\alpha_2}{R^2}; l_{20} = \frac{1}{R^2}; l_{21} = \frac{1}{R^4}; l_{21} = \frac{1}{R^4}$$
 (53g)

321 
$$l_{19} = \left[ -l_4 + \frac{1}{2} \left[ \frac{2\cos QT_0}{Q^2 R^2 [R^2 - Q^2]} + \frac{\cos[Q + R]T_0}{Q[RQ]^2 [2R + Q]} + \frac{\cos[Q - R]T_0}{Q[RQ]^2 [2R - Q]} \right] \right]$$
 (53h)

322 On substituting (53, b-d) on (53a), we have

323 
$$T_1 = \alpha_2 l_{14} + \bar{\xi}_1 l_{16} + T_0 l_{17} + l_{18}$$
 (54)

We, now, determine the maximum values of  $\varsigma(t)$  and  $\eta(t)$  say  $\varsigma_a$  and  $\eta_c$  respectively by

evaluating (46 a, b) at the critical values namely  $t=t_a, \tau=\tau_a$  and  $T=T_c, \tau=\tau_c$ .

326 
$$\zeta_a = \zeta^{(1)}(t_a, \tau_a)\varepsilon + \zeta^{(2)}(t_a, \tau_a)\varepsilon^2 + \dots$$
 (55a)

327 
$$\eta_c = \eta^{(1)}(T_c, \tau_c)\varepsilon + \eta^{(2)}(T_c, \tau_c)\varepsilon^2 + \dots$$
 (55b)

328 Expanding (55 a) in Taylor series using,

329 
$$t_a = t_0 + \varepsilon t_1 + \varepsilon^2 t_2 + \dots; \tau_a = \varepsilon t_a = \varepsilon \left[ t_0 + \varepsilon t_1 + \varepsilon^2 t_2 + \dots \right]$$
 (56a)

330 we have

331 
$$\zeta_a = \varepsilon \left[ \zeta^{(1)}(t_0, 0) + \zeta_{,t}^{(1)}(t_0, 0) \left[ \varepsilon t_1 + \varepsilon^2 t_2 + \ldots \right] + \zeta_{,\tau}^{(1)}(t_0, 0) \varepsilon \left[ t_0 + t_1 \varepsilon_1 + \ldots \right] \right]$$

332 
$$+ \varsigma^{(2)}(t_0,0)\varepsilon^2 + \dots$$
 (56b)

333 Regrouping the terms in orders of  $\mathcal{E}$  yields

334 
$$\zeta_a = \varepsilon \zeta^{(1)}(t_0, 0) + \varepsilon^2 \left[ t_1 \zeta_{,t}^{(1)}(t_0, 0) + t_0 \zeta_{,\tau}^{(1)}(t_0, 0) + \zeta^{(2)}(t_0, 0) \right] + \dots$$
 (56c)

On substituting the terms in (56c) from (51, b-d), we have

336 
$$\zeta_a = 2\bar{\xi}_1 l_{10}\varepsilon + \left[ t_0 \bar{\xi}_1 l_9 + 2\bar{\xi}_1 l_{11} + k_1 \bar{\xi}_1^2 l_{12} + k_2 \bar{\xi}_2^2 \left[ \frac{S}{R^2} \right]^2 l_{13} \right] \varepsilon^2 + \dots$$
 (57)

337 Similarly, expanding (55 b) in Taylor series using,

338 
$$T_c = T_0 + \varepsilon T_1 + \varepsilon^2 T_2 + ...; \tau_c = \varepsilon T_c = \varepsilon \left[ T_0 + \varepsilon T_1 + \varepsilon^2 T_2 + ... \right]$$
 (58a)

339 we have

340 
$$\eta_c = \varepsilon \left[ \eta^{(1)}(T_0, 0) + \eta_{,t}^{(1)}(T_0, 0) \left[ \varepsilon T_1 + \varepsilon^2 T_2 + \ldots \right] + \eta_{,\tau}^{(1)}(T_0, 0) \varepsilon \left[ T_0 + \varepsilon_1 T_1 + \ldots \right] \right]$$

341 
$$+\eta^{(2)}(T_0,0)\varepsilon^2+...$$
 (58b)

Regrouping the terms in orders of  $\mathcal{E}$  yields

343 
$$\eta_c = \varepsilon \eta^{(1)}(T_0, 0) + \varepsilon^2 \left[ T_1 \eta_{,\tau}^{(1)}(T_0, 0) + T_0 \eta_{,\tau}^{(1)}(T_0, 0) + \eta^{(2)}(T_0, 0) \right] + \dots$$
 (58c)

On substituting the terms in (58c) from (53, b-d), we have

345 
$$\eta_c = 2S\bar{\xi}_2 l_{20}\varepsilon + \left[ T_0 S^2 \bar{\xi}_2 l_{18} + S^2 \bar{\xi}_2 l_{21} + R^2 S\bar{\xi}_1 \bar{\xi}_2 l_{19} \right] \varepsilon^2 + \dots$$
 (59)

346 The net maximum displacement  $\xi_{\scriptscriptstyle m}$  is

347 
$$\xi_m = \zeta_a + \eta_c = \zeta(t_a, \tau_a) + \eta(T_c, \tau_c)$$
 (60)

348 Substituting for terms in (60) from (57) and (59) we get

349 
$$\xi_m = C_1 \varepsilon + C_2 \varepsilon^2 + \dots$$
 (61a)

350 where

351 
$$C_1 = l_{22}; C_2 = \bar{\xi}_1 l_{23} + S^2 \bar{\xi}_2 l_{24} + k_1 \bar{\xi}_1^2 l_{12} + k_2 \bar{\xi}_2^2 \left[ \frac{S}{R^2} \right]^2 l_{13} + R^2 \bar{\xi}_1 \bar{\xi}_2 S l_{19}$$
 (61b)

352 
$$l_{22} = 2\bar{\xi}_1 l_{10} + 2S\bar{\xi}_2 l_{20}; l_{23} = 2l_{11} + t_0 l_9; l_{24} = l_{21} + T_0 l_{18}$$
 (61c)

As noted by [1-3] and [21], the condition for dynamic buckling is

$$354 \qquad \frac{d\lambda}{d\xi_m} = 0 \tag{62}$$

As in [23-24], applying the method of reversal of series of (61a), we get

356 
$$\varepsilon = d_1 \xi_m + d_2 \xi_m^2 + \dots$$
 (63)

357 Substituting for  $\xi_m$  from (61a) in (63) and equating powers of orders of  $\varepsilon$  , we get

358 
$$d_1 = \frac{1}{C_1}, d_2 = -\frac{C_2}{C_1^3}$$
 (64)

- 359 The maximization in (62) is better done from (63), thus implementing (62) using (63) we
- 360 have

361 
$$\xi_m(\lambda_D) = \frac{C_1^2}{2C_2}$$
 (65)

- where,  $\xi_m(\lambda_D)$  is the value of the net displacement at buckling. In determining the dynamic
- 363 buckling load, we evaluate (63) at
- 364  $\lambda = \lambda_D$
- 365 to yield

373

375

366 
$$\varepsilon = \xi_m(\lambda_D) [d_1 + d_2 \xi_m]_{(\lambda = \lambda_D)}$$
 (66)

On substituting for terms  $d_1$  and  $d_2$  from (64) and  $\xi_m(\lambda_D)$  from (65) in (66) and simplify to get

$$\mathcal{E}\lambda_D = \frac{C_1}{4C_2} \tag{67}$$

369 The expansion of (67) gives [using (61b, c)]

370 
$$\lambda_{D} = \frac{1}{4} \left[ \frac{\omega_{0}}{\omega_{1}} \right]^{2} \left[ 2\bar{\xi}_{1} l_{10} + 2S\bar{\xi}_{2} l_{20} \right] \left[ \bar{\xi}_{1} l_{23} + S^{2}\bar{\xi}_{2} l_{24} + k_{1}\bar{\xi}_{1}^{2} l_{12} + k_{2}\bar{\xi}_{2}^{2} \left[ \frac{S}{R^{2}} \right]^{2} l_{13} \right]^{-1} + R^{2}\bar{\xi}_{1}\bar{\xi}_{2}S l_{19}$$

$$(68)$$

- 371 Here, (68) gives the formula for evaluating the dynamic buckling load  $\lambda_{\!\scriptscriptstyle D}$  , and is valid for
- 372  $R \neq (1,2,Q,2Q,1-Q,1+Q)$  and  $Q \neq (R,2R,1-R,1+R,0,2R-1)$

# 374 4. ANALYSIS OF RESULT.

- We note that the results display all the imperfection parameters stated in problems (3)-
- 377 (5). This is unlike Danielson's problem in which the axisymmetric imperfection was neglected
- 378 for easy solution. In fact, the method is such that we can adequately account for all modal
- imperfections allowed in the formulation .The contributions of the quadratic terms  $k_1\xi_1^2, k_2\xi_2^2$

and the coupling term  $\xi_1 \xi_2$  are respectively given in the denominator of (68) by

381 
$$k_1 \bar{\xi}_1^2 l_{11}, k_2 \bar{\xi}_2^2 \left[ \frac{S}{R^2} \right] l_{13} \text{ and } R^2 \bar{\xi}_1 \bar{\xi}_2 S l_{19}.$$

- Thus if we assume that the axysymmetric imperfections are zero then  $\bar{\xi}_1 = 0$ , and the
- dynamic buckling load  $\lambda_D$  responsible for the buckling in this case is obtained from (68) as

384 
$$\lambda_{D} = \frac{1}{4} \left[ \frac{\omega_{0}}{\omega_{1}} \right]^{2} \left[ 2S \, \bar{\xi}_{2} \, l_{20} \right] \left[ S^{2} \, \bar{\xi}_{2} \, l_{24} + k_{2} \, \bar{\xi}_{2}^{2} \left[ \frac{S}{R^{2}} \right]^{2} l_{13} \right]^{-1} \right]$$
 (69)

- We note from (69), that, the effects of the coupling terms  $\xi_1 \xi_2$ ,  $\xi_1 \xi_0$  and the quadratic term
- 386  $k_1 \xi_1^2$  are zeros. The effect of the quadratic term  $k_2 \xi_2^2$  is non-zero and it is this term that
- dominates the buckling process. Neglecting  $\xi_1$  is sufficient to completely nullify the effect of
- 388  $\xi_1^2$  where the converse is not necessarily the case.
- However, if the non-axymmetric imperfections are neglected then  $\bar{\xi}_2 = 0$ , and the dynamic
- 390 buckling load  $\lambda_D$  following (68) become

391 
$$\lambda_{D} = \frac{1}{4} \left[ \frac{\omega_{0}}{\omega_{1}} \right]^{2} \left[ 2\bar{\xi}_{1} l_{10} \right] \left[ \bar{\xi}_{1} l_{23} + k_{1} \bar{\xi}_{1}^{2} l_{12} \right]^{-1}$$
 (70)

- We deduce from (70), that, the effects of the coupling terms  $\xi_1 \xi_2$ ,  $\xi_2 \xi_0$  and the quadratic
- term  $k_2 \xi_2^2$  are again zeros. The effect of the quadratic term  $k_1 \xi_1^2$  is non-zero and this
- singular term is the only non-linear term that influences the buckling process. Neglecting  $\bar{\xi}_2$
- is sufficient to completely nullify the effect of  $\xi_2^2$  where the converse is not necessarily the
- 396 case.
- 397 The results also confirm that the only condition under which the effects of the coupling term
- 398  $\bar{\xi}_1 \bar{\xi}_2$  would be felt is if none of the imperfection parameters in the shape of the mode
- coupling is neglected. In other word, is that nether the imperfection parameter  $\bar{\xi}_1$  nor  $\bar{\xi}_2$
- 400 should vanish for post dynamic buckling behavior of the structures. Once an imperfection is
- 401 neglected the coupling effect of the mode that is in the shape of the neglected imperfection,
- 402 with any other mode is neglected.

The graphical view of this phenomenon, we assume the following values.  $k_1=0.2,\ k_2=0.3,\ \bar{\xi}_1=0.02,\ \bar{\xi}_2=0.03,\ \alpha_1=0.01$  and  $\alpha_2=0.03$ . By varying  $\bar{\xi}_2$  and  $\alpha_2$  while keeping  $\bar{\xi}_1$  constant at 0.02 and  $\alpha_1=0$ , the corresponding values of  $\lambda_D$  were computed from (68). The plots of dynamic buckling load against the imperfection parameter and light viscous damping of the discretized spherical cap are shown in figures 1 and 2 below.

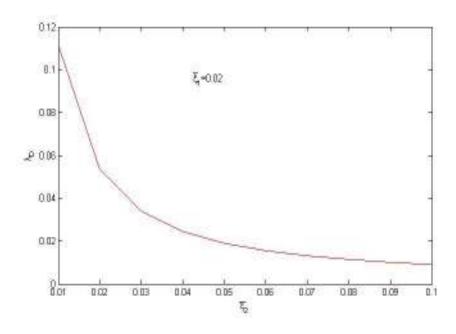


Figure 1: Dynamic buckling load of a spherical cap against the imperfection parameter  $\,\bar{\xi}_2$  (  $\,\bar{\xi}_1=0.02$  )

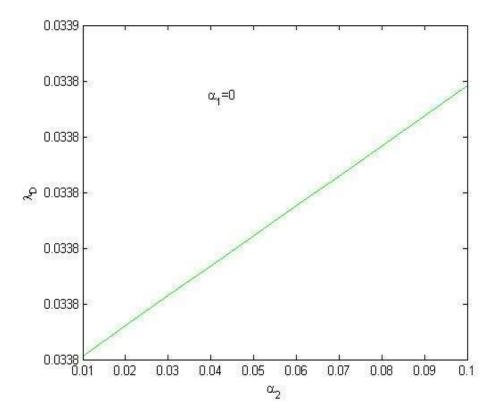


Figure 2: Dynamic buckling load of a spherical cap against the light viscous damping  $\alpha_2(\alpha_1 = 0)$ 

From fig.2 above we observe that dynamic buckling load increases with increased damping. Also in fig.1 dynamic buckling load increases if the structure is less imperfect, in other word, dynamic buckling load decreases with increased imperfection.

### 5. CONCLUSION.

From the above discussions, we note that while neglecting the imperfection parameters  $\bar{\xi}_1$  and  $\bar{\xi}_2$  automatically implies, among other things, neglecting the effects of the non-linear terms  $k_1\xi_1^2$  and  $k_2\xi_2^2$  respectively. Also, we observe that the only condition under which the effect of the coupling term  $\xi_1\xi_2$  would be felt, is when the imperfection parameters  $\bar{\xi}_1$  and  $\bar{\xi}_2$  are not equal to zero. Moreover, our results confirm those obtained by [21]. Finally, we

428	notice that we can determine the value of the dynamic buckling load $\lambda_{\!\scriptscriptstyle D}$ for whatever
429	number of modal imperfections.
430	
431	REFERENCES
432	[1] Hutchinson, J.W and Budiansky, B., Dynamic buckling estimates, AIAA J. 1966; 4: 525-
	<u> </u>
433	<b>530.</b>
434	[2] Budiansky, B. and Hutchinson, J.W., <i>Dynamic</i> buckling of imperfection- sensitive
435	structure, proceedings of the XIth Inter. Congr. Applied Mech., Springer-Verlag, Berling. 1966
436	[3] Budiansky, B., Dynamic buckling of elastic structures; Criteria and estimates, in, Dynamic
437	stability of structures.Pergamons, New York. 1966
438	[4] Ohsaki, M., Imperfection sensitivity of optimal, symmetric braced frames against
439	buckling. Int. J.Non-Linear Mechanics, 2003; 38(7):1103-1117.
440	[5] Dumir, P.C., Dube, G.P. and Mullick, A. Axisymmetric static and dynamic buckling of
441	laminated thick truncated conical cap. Int. J. Non-Linear Mechanics, 2003; 38(6):903-910.
442	[6] Kardomateas, G.A., Simites, G.J., Shem, L.and Li, R.,
443	Buckling of sandwich wide columns, Int. J. Non-Linear Mechanics, 2002; 37(7): 1239-1247.
444	[7] Anwen, W.and Wenying, Characteristic-value analysis for plastic dynamic buckling of
445	columns under elastoplastic compression waves, Int. J. Non-Linear Mechanics, 2003; 38(5):
446	615-628.
447	[8] Rafteyiannis, I.G. and Kounadis, A., Interaction under step loading, Int. J. Non-Linear
448	Mechanics, 2000; 35(3): 531-542.
0	
449	[9] Wei, Z.G., Yu, J.L. and Batra, R.C., Dynamic buckling of thin cylindrical shells under axial
450	impact, Int.J.Impact Engng., 2005; 32:572-592.
451	[10] Batra, R.C. and Wei, Z.G. (2005), Dynamic buckling of thin thermoviscoplastic
452	rectangular plate, J.of Thin-Walled Structures, 2005; 43: 273-290.
453	[11] Zhang, T., Liu, T.G., Zhao, Y. and Luo J.Z., Nonlinear dynamic buckling of stiffened

plates under in-plane impact load. J. of Zhejiang University of Science, 2004; 5 (5): 609-617

455 456	[12] Danielson, D., Dynamic buckling loads of imperfection sensitive structures from perturbation procedures, AIAAJ.1969; 7:1506-1510
457	[13] Aksogan O. and Sofiyev, A.V., Dynamic buckling of cylindrical shells with variable
458	thickness subjected to a time-dependent external pressure varying as a power function of
	, , , , , , , , , , , , , , , , , , , ,
459	time, J. of Sound and vibration, 2002; 254(4): 693-703.
460	[14] Schenk, C.A and Schueller, G.I., Buckling of cylindrical shells with random
461	imperfections, Int. J. Non-Linear mechanics, 2003; 38: 1119-1132.

- [15] Wang, A. and Tian, W.Twin characteristics- parameter solution under elastic compression waves, Int J. Solids Structures, 2002; 39:861-877.
- [16] Wang, A. and Tian, W.Twin characteristic parameter solutions of axisymmetric dynamic
   plastic buckling for cylindrical shells under axial compression waves, Int. J. solids
- 466 Structures, 2003; 40:3157-3175
- [17] Ette, A.M.On a two-small-parameter dynamic buckling of a lightly damped spherical cap trapped by a step load, Journal of the Nigerian Mathematics Society, 2007; 23:7-26.

[18] Ette, A.M. On a two-small-parameter dynamic stability of a lightly damped spherical

- shell pressurized by a harmonic excitation, Journal of the Nigerian Association of
- 471 Mathematical Physics, 2007; 11:333-362
- 472 [19] Ette, A.M.On the buckling of lightly damped cylindrical shells modulated by a periodic
- load, Journal of the Nigerian Association of Mathematical Physics, 2006; 10: 327-344
- 474 [20] Ette, A.M.Perturbation technique on the dynamic stability of a damped cylindrical shell
- 475 axially stressed by an impulse, Journal of the Nigerian Association of Mathematical Physics,
- 476 2008; 12:103-120.

- 477 [21] Ette, A.M. Dynamic buckling of imperfect spherical shell under an axial Impulse,
- 478 Int.J.Non-Linear Mechanics, 1997; 32(1):201-209
- 479 [22] Ette, A.M. Perturbation approached on the dynamic buckling of a lightly damped
- 480 cylindrical shells modulated by a periodic load, Journal of the Nigerian Mathematics Society,
- 481 2009; 28: 97-135.

- 482 [23] Amazigo, J.C.Buckling of stochastically imperfect columns of nonlinear elastic
- foundations, Quart.App.Math.1973; 31:403.
- 484 [24] Amazigo, J.C. Dynamic buckling of structures with random imperfections. Stochastic
- 485 problem in Mechanics. Ed. H. Leipolz, University of Waterloo press, 1974; 243-254.

486 487

#### **APPENDIX**

```
488
               k1=0.2;
489
               k2=0.3;
490
               zi2bar=0.03;
491
               zi1bar=0.02;
492
              alpha1=0.01;
493
               alpha2=0.03;
494
              R=0.1;
495
               Q = 0.3;
496
              S=(R/Q)^2;
497
              L0 = -1/Q^4;
498
              L1=-1/(2*Q^2) + 1/(6*Q^2);
499
               L2=0.5 + 1/(2*(Q^2-4*R^2))-2/(R^2*(Q^2-4*R^2));
500
              L3=-1/R^4;
501
               L4=1/(R^2*Q)^2 + 0.5*(-2/((R*Q)^2*(R^2-Q^2))-1/(Q*(R*Q)^2*(2*R+Q)) + (R^2-Q^2)
502
               1/(Q*(R*Q)^2*(2*R-Q));
503
               L5=1/Q^2;
504
               t0=pi/Q;
505
               Bt0=pi/R;
506
               phi=1/(2*Q)*(1+2*k1*zi1bar);
               L6=sin(2*Q*t0)/(3*Q^2);
507
508
               omega=1/(2*R)*(S-R^2*zi1bar/Q^2);
509
               L7=Q^2*\sin(2*R*t0)/(R^3*(Q^2-4*R^2));
510
               L8=phi/Q;
511
               L9=-alpha1/Q^2;
512
              L10=1/Q^2;
513
               L11=1/0^4;
514
               L12=1/Q^2-1/(3*Q^4);
515
               L13=-1 -1/(2*(Q^2-4*R^2)) + 1/(R^2*(Q^2-4*R^2))-Q^2*(1/(Q^2-4*R^2))+
516
                2/(Q^2-R^2);
517
               L14=-cos(R*Bt0)/R^2;
518
               L15=-R*L3*sin(R*Bt0);
519
              L16 = -R^3 * S * L4 * sin(R * Bt0) - (R^2/2) * (2 * sin(Q * Bt0) / (Q * R^2 * (R^2 - Q^2)) -
520
               \cos(0*Bt0)/((R*0)^2*(2*R+0))-(0+R)*\sin(0+R)*Bt0/(0*(R*0)^2*(2*R+0))-
521
               (Q-R)*sin(Q-R)*Bt0/(Q*(R*Q)^2*(2*R-Q)));
522
               L17=omega/R;
523
               L18=-alpha2/R^2;
524
               L19=-L4 + 0.5*(2*cos(Q*Bt0)/(Q^2*R^2*(R^2-Q^2)) +
525
               \cos(Q+R)*Bt0/(Q*(R*Q)^2*(2*R+Q)) + \cos(Q-R)*Bt0/(Q*(R*Q)^2*(2*R+Q)) + \cos(Q-R)*(Q*(R*Q)^2*(2*R+Q)^2*(2*R+Q)) + \cos(Q-R)*(Q*(R*Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*(2*R+Q)^2*
526
               Q)));
527
               L20=1/R^2;
528
               L21=1/R<sup>4</sup>;
529
               L22=2*zi1bar*L10 + 2*S*zi2bar*L20;
530
               L23=2*L11 + t0*L9;
531
               L24=L21+Bt0*L18;
```

```
532
        c1=L22;
533
        c2=zi1bar*L23+S^2*zi2bar*L24+k1*zi1bar^2*L12 +
        k2*zi2bar^2*(S/R^2)^2*L13+R^2*zi1bar*zi2bar*S*L19;
534
535
536
        LambdaD=c1/(4*c2*Q^2);
        z=LambdaD;
537
538
        INTERPRETATION OF VARIBLES
539
        {\rm zi1bar}=\bar{\xi}_1 \text{ , zi2bar}=\bar{\xi}_2 \text{ , alpha1}=\alpha_1 \text{ , alpha2}=\alpha_2 \text{ , t0}=t_0 \text{ , Bt0}=T_0 \text{ , phi}=\varphi \text{ ,}
540
        Omega = \Phi , pi= \pi , LambdaD = \lambda_D
541
542
```